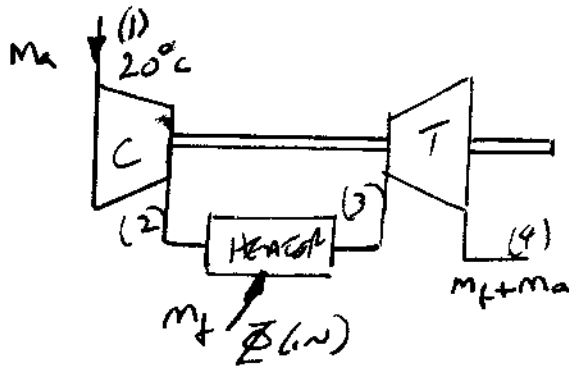


Ø 2000



$$\Gamma_p = 10 \quad T_1 = 293 \text{ K}$$

$$T_3 = 1173 \text{ K}$$

GAS CONSTANTS AIR

$$R_a = R_0 / \bar{m} = 287 \text{ J/kgK}$$

$$C_{pa} = 1.004 \text{ kJ/kgK}$$

$$\gamma_a = 1.4$$

GAS CONSTANTS (GASOLINE)

$$R_g = R_0 / 32 = 259.81 \text{ J/kgK}$$

$$C_{pg} = 1.2 \text{ kJ/kgK}$$

$$C_{vg} = C_{pg} - R_g = 1200 - 259.81 = 940.2 \text{ J/kgK}$$

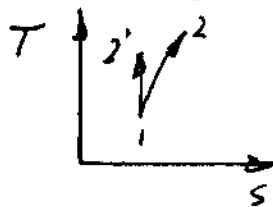
$$\gamma_g = C_{pg} / C_{vg} = 1200 / 940.2 = 1.276$$

COMPRESSOR

$$T_2' = 293 \times 10^{\frac{\Gamma_p - 1}{\gamma_a}}$$

$$T_2 = 293 \times 10^{0.286} = 566 \text{ K}$$

ISENTROPIC EFFICIENCY



$$\eta_{is} = \frac{T_2' - T_1}{T_2 - T_1} = 0.88 = \frac{566 - 293}{T_2 - 293}$$

$$T_2 = 603.2 \text{ K}$$

$$\Phi(\text{in}) = m_g C_{pg} T_3 - m_a C_{pa} T_2$$

USING FUEL CALORIFIC VALUE

$$\Phi(\text{in}) = m_f \times 45000 \quad m_g = m_a + m_f$$

$$45000 m_f = (m_a + m_f) \times 1.2 \times 1173 - m_a \times 1.004 \times 603.2$$

$$45000 = \left(\frac{m_a}{m_f} + 1 \right) \times 1.2 \times 1173 - \frac{m_a}{m_f} \times 1.004 \times 603.2$$

$$45000 = 1407.6 m_a/m_f + 1407.6 - 605.5 m_a/m_f$$

$$43592 = 802 m_a/m_f \quad m_a/m_f = \underline{\underline{54.35}}$$

Q6 2000

TURBINE $p_2 = 10 \times 1 = 10 \text{ bar}$
 $p_3 = 90\% \times p_2 = 9 \text{ bar}$ $\gamma_p = 9$
 $T_4' = T_3 \left(\frac{1}{\gamma_p}\right)^{\frac{\gamma-1}{\gamma}} = 1173 \left(\frac{1}{9}\right)^{-226/1173} = 729 \text{ K}$

ISOTHERMAL EFFICIENCY

$$\eta_{is} = \frac{T_3 - T_4}{T_3 - T_4'} \quad 0.9 = \frac{1173 - T_4}{1173 - 729}$$



$$T_4 = 773.6 \text{ K}$$

$$\text{Power out} = \dot{m}_g C_{pg} (T_3 - T_4)$$

$$P(\text{out}) = \dot{m}_g \times 1.2 \times (1173 - 773.6) = 479.2 \dot{m}_g$$

COMPRESSOR

$$\begin{aligned} P(\text{in}) &= \dot{m}_a C_{pa} (T_2 - T_1) \\ &= \dot{m}_a \times 1.004 (603.2 - 293) \\ &= 311.4 \dot{m}_a \end{aligned}$$

NET POWER

$$P_{\text{net}} = 479.2 \dot{m}_g - 311.4 \dot{m}_a$$

$$\dot{Q}(\text{in}) = 45000 \dot{m}_f$$

EFFICIENCY

$$\eta_{th} = \frac{P_{\text{net}}}{\dot{Q}_{\text{in}}} = \frac{479.2 \dot{m}_g - 311.4 \dot{m}_a}{45000 \dot{m}_f}$$

$$\dot{m}_g = \dot{m}_a + \dot{m}_f \quad \dot{m}_a = 54.35 \dot{m}_f \quad \dot{m}_g = 55.35 \dot{m}_f$$

$$\eta_{th} = \frac{479.2 \times 55.35 \dot{m}_f - 311.4 \times 54.35 \dot{m}_f}{45000 \dot{m}_f}$$

$$\eta_{th} = \frac{9612}{45000} = 0.214 \text{ or } 21.4\%$$

Q6 2500

CARNOT EFFICIENCY

$$\eta_c = 1 - \frac{T_{\text{COLD}}}{T_{\text{HOT}}}$$

$$= 1 - \frac{293}{1173} = 0.75 \text{ or } 75\%$$